

SHAPE DESIGN SENSITIVITY FORMULATION FOR FINITE DEFORMATION ELASTO- PLASTICITY WITH CONTACT

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INTRODUCTION

⇒ Finite Deformation Elastoplasticity

- Multiplicative Decomposition of \mathbf{F}
- Stress-Free Intermediate Configuration
- Return Mapping in Principal Stress Space
- Exact Tangent Operator

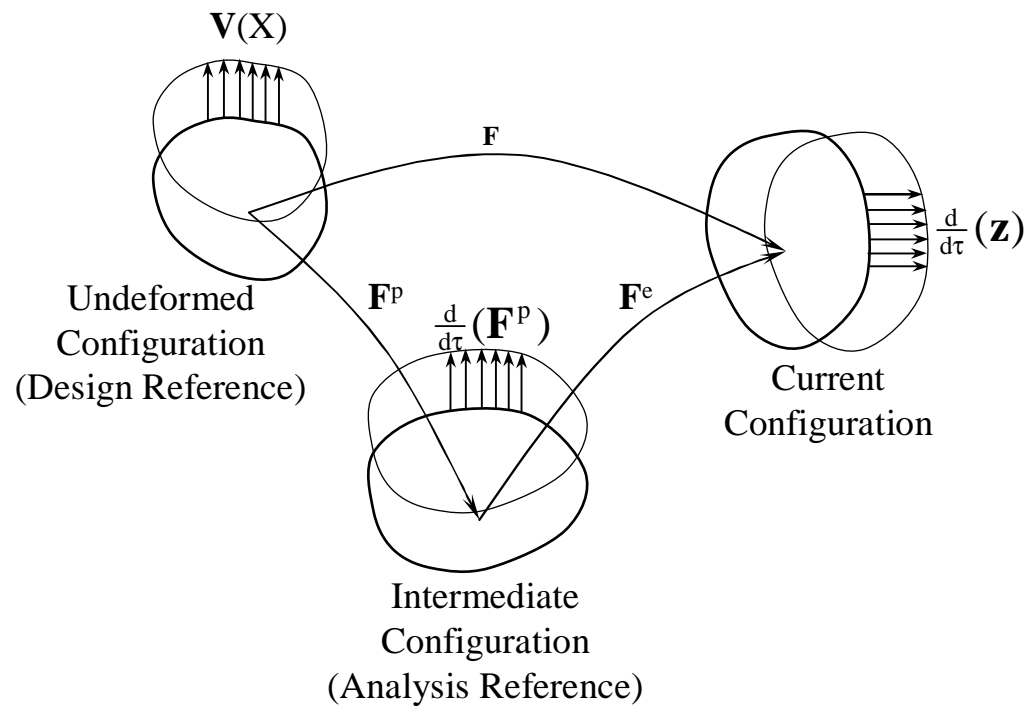
⇒ Frictional Contact Problem

- Continuum-Based Contact Formulation
- Penalty Regularization
- Regularized Coulomb Friction Model

⇒ Meshfree Discretization

- Reproducing Kernel Particle Method (RKPM)
- Direct Transformation Method for Essential B. C.

ANALYSIS AND DESIGN PROCEDURE



VARIATIONAL FORMULATION

Governing Equation & Linearization

$$a_{\Omega}({}^{n+1}\mathbf{z}, \bar{\mathbf{z}}) = \ell_{\Omega}(\bar{\mathbf{z}}), \quad \forall \bar{\mathbf{z}} \in Z \quad a_{\Omega}(\mathbf{z}, \bar{\mathbf{z}}) = \int_{\Omega} \boldsymbol{\tau} : \bar{\boldsymbol{\varepsilon}} \, d\Omega$$

$$a_{\Omega}^*({}^{n+1}\mathbf{z}^k; \Delta \mathbf{z}^{k+1}, \bar{\mathbf{z}}) = \ell_{\Omega}(\bar{\mathbf{z}}) - a_{\Omega}({}^{n+1}\mathbf{z}^k, \bar{\mathbf{z}}) \quad \forall \bar{\mathbf{z}} \in Z$$

$$a_{\Omega}^*({}^{n+1}\mathbf{z}^k; \Delta \mathbf{z}^{k+1}, \bar{\mathbf{z}}) = \int_{\Omega} [\bar{\boldsymbol{\varepsilon}} : \mathbf{c} : \boldsymbol{\varepsilon}(\Delta \mathbf{z}^{k+1}) + {}^{n+1}\boldsymbol{\tau}^k : \boldsymbol{\eta}(\Delta \mathbf{z}^{k+1}, \bar{\mathbf{z}})] \, d\Omega$$

Consistent Tangent Operator

$$\mathbf{c} = \frac{\partial \boldsymbol{\tau}}{\partial \boldsymbol{\varepsilon}} = \sum_{i=1}^3 \sum_{j=1}^3 \mathbf{c}_{ij}^{\text{alg}} \mathbf{m}^i \otimes \mathbf{m}^j + 2 \sum_{i=1}^3 \tau_i^p \hat{\mathbf{c}}^i$$

$$\mathbf{c}^{\text{alg}} \equiv \frac{\partial \boldsymbol{\tau}^p}{\partial \mathbf{e}^{\text{tr}}} = \mathbf{c}^e - 4\mu^2 \mathbf{A} \mathbf{N} \otimes \mathbf{N} - \frac{4\mu^2 \hat{\gamma}}{\|\boldsymbol{\eta}^{\text{tr}}\|} [\mathbf{I}_{\text{dev}} - \mathbf{N} \otimes \mathbf{N}]$$

DESIGN SENSITIVITY ANALYSIS

DSA Equation

$$\frac{d}{dt} [a_{\Omega}^{(n+1)}(\mathbf{z}, \bar{\mathbf{z}})] = \frac{d}{dt} [\ell_{\Omega}(\bar{\mathbf{z}})], \quad \forall \bar{\mathbf{z}} \in Z$$

$$a_{\Omega}^*(\mathbf{z}; \dot{\mathbf{z}}, \bar{\mathbf{z}}) = \ell'_{\mathbf{v}}(\bar{\mathbf{z}}) - a'_{\mathbf{v}}(\mathbf{z}, \bar{\mathbf{z}}), \quad \forall \bar{\mathbf{z}} \in Z$$

Fictitious Load

$$a'_{\mathbf{v}}(\mathbf{z}, \bar{\mathbf{z}}) = \int_{\Omega} (\bar{\boldsymbol{\varepsilon}} : \mathbf{c} : \boldsymbol{\varepsilon}_{\mathbf{v}}(\mathbf{z}) + \bar{\boldsymbol{\varepsilon}} : \mathbf{c} : \boldsymbol{\varepsilon}_{\mathbf{p}}(\mathbf{z}) + \boldsymbol{\tau}^{\text{fic}} : \bar{\boldsymbol{\varepsilon}}) d\Omega \\ + \int_{\Omega} (\boldsymbol{\tau} : \boldsymbol{\eta}_{\mathbf{v}}(\mathbf{z}, \bar{\mathbf{z}}) + \boldsymbol{\tau} : \boldsymbol{\eta}_{\mathbf{p}}(\mathbf{z}, \bar{\mathbf{z}}) + \boldsymbol{\tau} : \bar{\boldsymbol{\varepsilon}} \text{div} \mathbf{V}) d\Omega$$

Design velocity dependent terms

$$\boldsymbol{\varepsilon}_{\mathbf{v}}(\mathbf{z}) = -\text{sym}(\nabla_0 \mathbf{z} \nabla_n \mathbf{V})$$

$$\boldsymbol{\eta}_{\mathbf{v}}(\mathbf{z}, \bar{\mathbf{z}}) = -\text{sym}(\nabla_n \bar{\mathbf{z}}^T \nabla_0 \mathbf{z} \nabla_n \mathbf{V}) - \text{sym}(\nabla_0 \bar{\mathbf{z}} \nabla_n \mathbf{V})$$

DESIGN SENSITIVITY ANALYSIS (cont.)

Path-dependent terms

$$\boldsymbol{\varepsilon}_p(\mathbf{z}) = -\text{sym}(\mathbf{G}) \quad \mathbf{G} = \mathbf{F}^e \frac{d}{d\tau}(\mathbf{F}^p) \mathbf{F}^{-1}$$

$$\boldsymbol{\eta}_p(\mathbf{z}, \bar{\mathbf{z}}) = -\text{sym}(\nabla_n \bar{\mathbf{z}}^T \mathbf{G})$$

$$\boldsymbol{\tau}^{\text{fic}} = \sum_{i=1}^3 \left[\frac{\partial \tau_i^p}{\partial \boldsymbol{\alpha}} \frac{d}{d\tau}(\boldsymbol{\alpha}_n) + \frac{\partial \tau_i^p}{\partial \hat{\mathbf{e}}^p} \frac{d}{d\tau}(\hat{\mathbf{e}}_n^p) \right] \mathbf{m}^i$$

Updating DSA Variables

$$\frac{d}{d\tau}(\boldsymbol{\alpha}_{n+1}) = \frac{d}{d\tau}(\boldsymbol{\alpha}_n) + \left(\mathbf{H}_\alpha + \sqrt{\frac{2}{3}} \mathbf{H}'_\alpha \hat{\gamma} \right) \frac{d}{d\tau}(\hat{\gamma}) \mathbf{N} + \mathbf{H}_\alpha \hat{\gamma} \frac{d}{d\tau}(\mathbf{N})$$

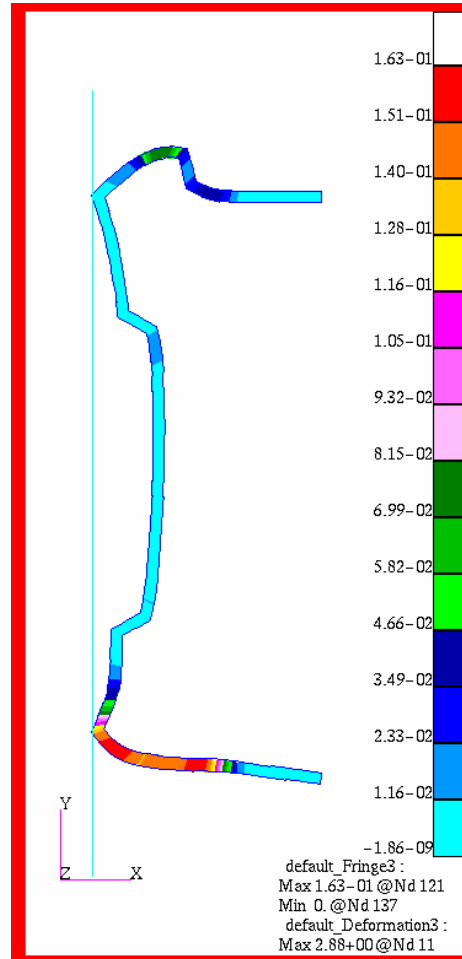
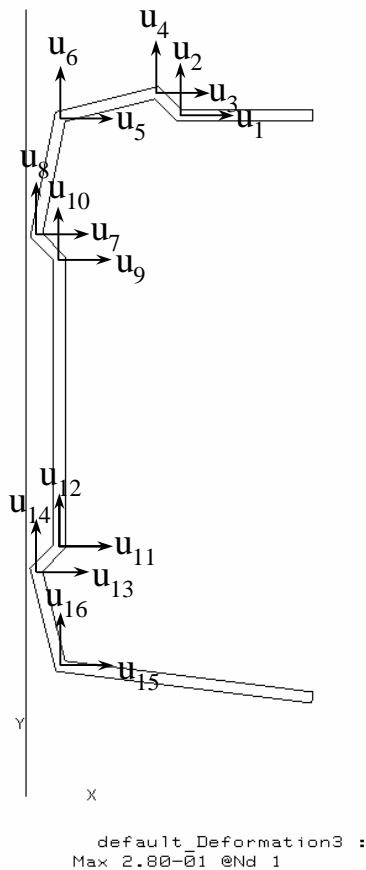
$$\frac{d}{d\tau}(\hat{\mathbf{e}}_{n+1}^p) = \frac{d}{d\tau}(\hat{\mathbf{e}}_n^p) + \sqrt{\frac{2}{3}} \frac{d}{d\tau}(\hat{\gamma})$$

$$\frac{d}{d\tau}(\mathbf{F}_{n+1}^p) = \frac{d}{d\tau}(\mathbf{F}_{n+1}^{e-1}) \mathbf{F}_{n+1} + \mathbf{F}_{n+1}^{e-1} \frac{d}{d\tau}(\mathbf{F}_{n+1})$$

$$\frac{d}{d\tau}(\mathbf{F}_{n+1}^e) = \frac{d}{d\tau}(\mathbf{f}^p) \mathbf{F}_{n+1}^{e \text{ tr}} + \mathbf{f}^p \frac{d}{d\tau}(\mathbf{F}_{n+1}^{e \text{ tr}})$$

$$\mathbf{f}^p = \sum_{j=1}^3 \exp(-\hat{\gamma} \mathbf{N}_j) \mathbf{m}^j \quad \begin{array}{l} \text{Incremental Plastic} \\ \text{Deformation Gradient} \end{array}$$

BUMPER IMPACT PROBLEM



Density	$\rho = 7,800 \text{ kg/m}^3$
Initial Velocity	$v_0 = 10 \text{ km/hr}$
Analysis Time	$t = 0 \sim 10 \text{ msec}$
Time Increment	$\Delta t = 0.1 \text{ msec}$
Mounting Displ.	$d = 2.8 \text{ cm}$
Thickness	$h = 0.5 \text{ cm}$
Contact Penalty No.	$w_n = 1,000$
Friction Coeff.	$\mu_f = 0.4$
Lame's Constants	$\lambda = 110.8 \text{ GPa}$ $\mu = 80.2 \text{ GPa}$
Plastic Hardening	$H = 1.1 \text{ GPa}$
Isotropic Hardening	
Initial Yield Stress	$\sigma_Y = 500 \text{ MPa}$
Newmark Parameters	$\gamma = 0.26$ $\beta = 0.5$

SENSITIVITY ANALYSIS RESULTS

Response Analysis
275 sec

Sensitivity Analysis
116 / 16 sec

Performance (Ψ)	$\Delta\Psi$	Ψ'	$(\Delta\Psi/\Psi')\times 100\%$	
u ₂				
\hat{e}^p_{15}	.680005E-01	-.179756E-07	-.179757E-07	100.00
\hat{e}^p_{65}	.164338E+00	.311392E-08	.311393E-08	100.00
\hat{e}^p_{29}	.126643E-01	-.901637E-10	-.901545E-10	
Z _{x39}	.429139E+00	.120943E-06	.120940E-06	100.00
F _{Cx100}	.379375E+01	.473864E-07	.473865E-07	100.00
u ₄				
\hat{e}^p_{15}	.680005E-01	.246181E-07	.246181E-07	100.00
\hat{e}^p_{65}	.164338E+00	.105172E-08	.105173E-08	100.00
\hat{e}^p_{29}	.126643E-01	.589794E-09	.589795E-09	100.00
Z _{x39}	.429139E+00	-.295825E-06	-.295824E-06	100.00
F _{Cx100}	.379375E+01	.335517E-09	.335511E-09	100.00
u ₆				
\hat{e}^p_{15}	.680005E-01	-.170857E-07	-.170857E-07	100.00
\hat{e}^p_{65}	.164338E+00	-.237257E-08	-.237256E-08	100.00
\hat{e}^p_{29}	.126643E-01	-.720239E-10	-.720198E-10	
Z _{x39}	.429139E+00	.167699E-06	.167698E-06	100.00
F _{Cx100}	.379375E+01	-.176290E-07	-.176292E-07	100.00
u ₈				
\hat{e}^p_{15}	.680005E-01	.581799E-09	.581877E-09	99.99
\hat{e}^p_{65}	.164338E+00	-.635253E-09	-.635254E-09	100.00
\hat{e}^p_{29}	.126643E-01	-.185890E-08	-.185890E-08	100.00
Z _{x39}	.429139E+00	-.397143E-07	-.397141E-07	100.00
F _{Cx100}	.379375E+01	.250196E-07	.250194E-07	100.00
u ₁₀				
\hat{e}^p_{15}	.680005E-01	-.262956E-09	-.262932E-09	100.01
\hat{e}^p_{65}	.164338E+00	.136684E-09	.136687E-09	100.00
\hat{e}^p_{29}	.126643E-01	.873228E-09	.873231E-09	100.00
Z _{x39}	.429139E+00	-.168128E-06	-.168129E-06	100.00
F _{Cx100}	.379375E+01	-.431408E-07	-.431402E-07	100.00
u ₁₂				
\hat{e}^p_{15}	.680005E-01	-.186975E-09	-.186998E-09	99.99
\hat{e}^p_{65}	.164338E+00	-.129114E-08	-.129115E-08	100.00
\hat{e}^p_{29}	.126643E-01	-.311886E-09	-.311878E-09	100.00
Z _{x39}	.429139E+00	-.304683E-07	-.304689E-07	100.00
F _{Cx100}	.379375E+01	.490867E-07	.490870E-07	100.00

Design Optimization
Problem Definition

MIN Area

S.T. $\hat{e}^p_{16}(0.10) \leq 0.05$

$\hat{e}^p_{62}(0.15) \leq 0.05$

$\hat{e}^p_{65}(0.16) \leq 0.05$

$\hat{e}^p_{66}(0.16) \leq 0.05$

$\hat{e}^p_{67}(0.15) \leq 0.05$

$\hat{e}^p_{60}(0.15) \leq 0.05$

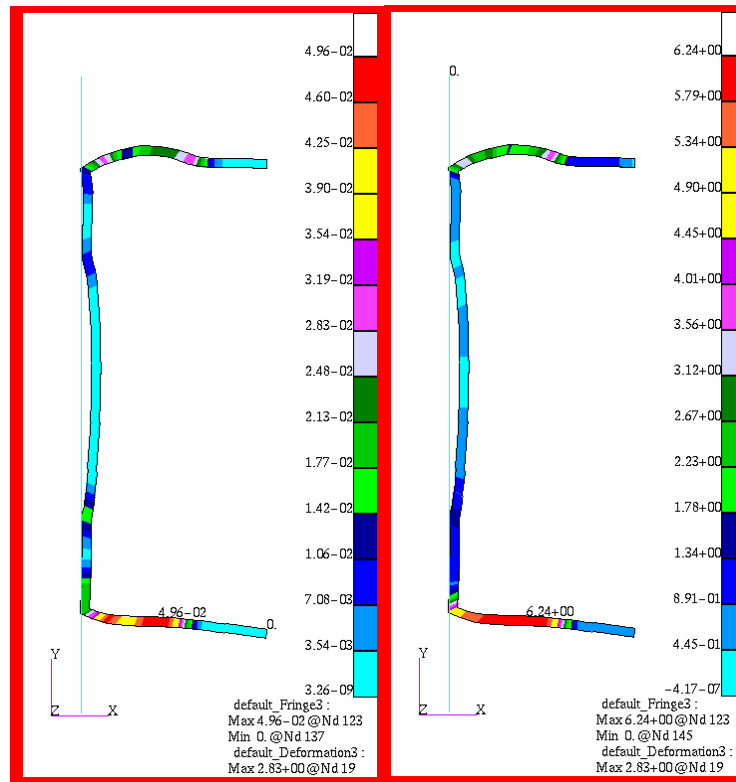
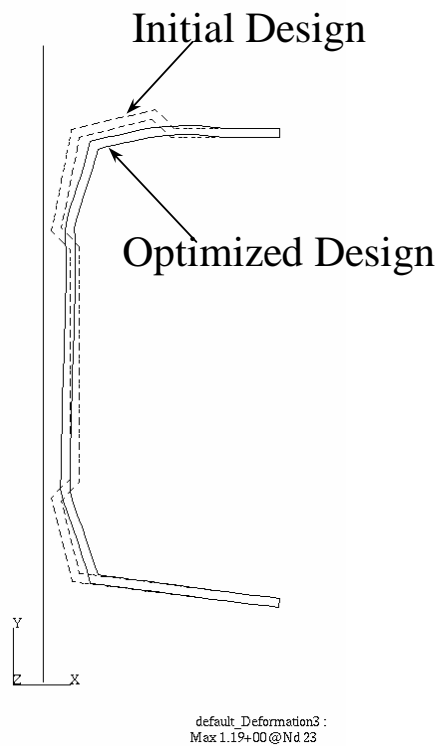
$\hat{e}^p_{61}(0.14) \leq 0.05$

$F_{Cx}(4.55) \geq 4.55$

$-1.0 \leq u_i \leq 1.0$

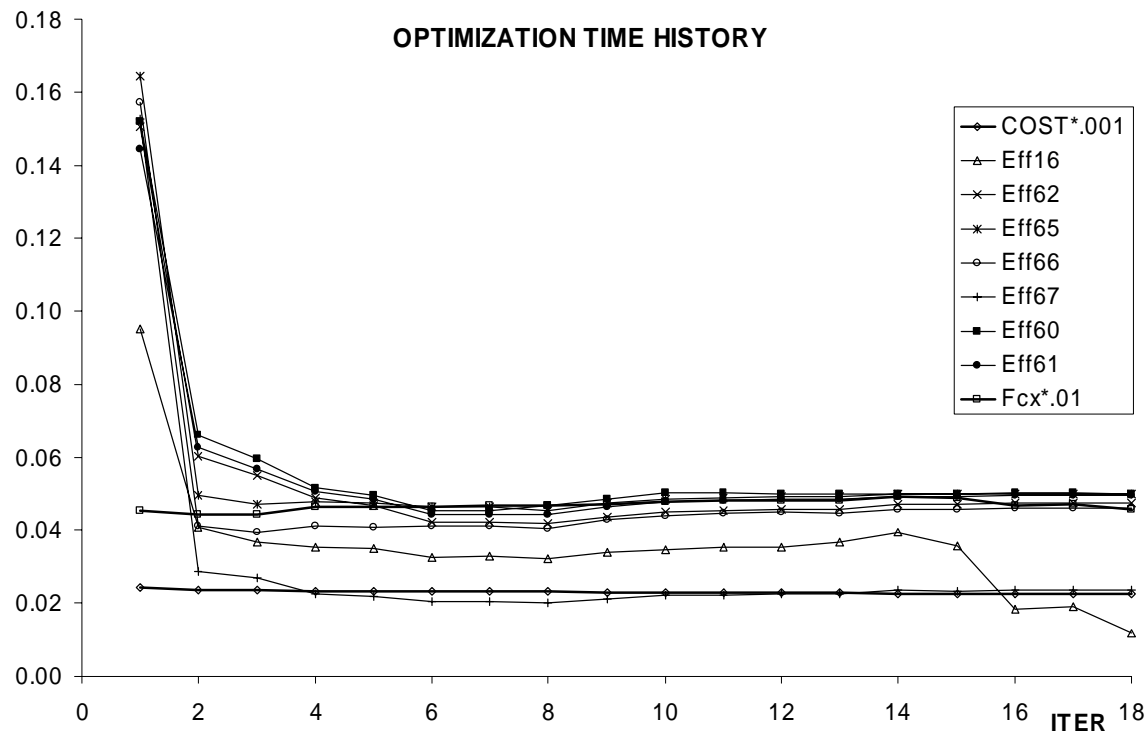
$i = 1,16$

DESIGN OPTIMIZATION



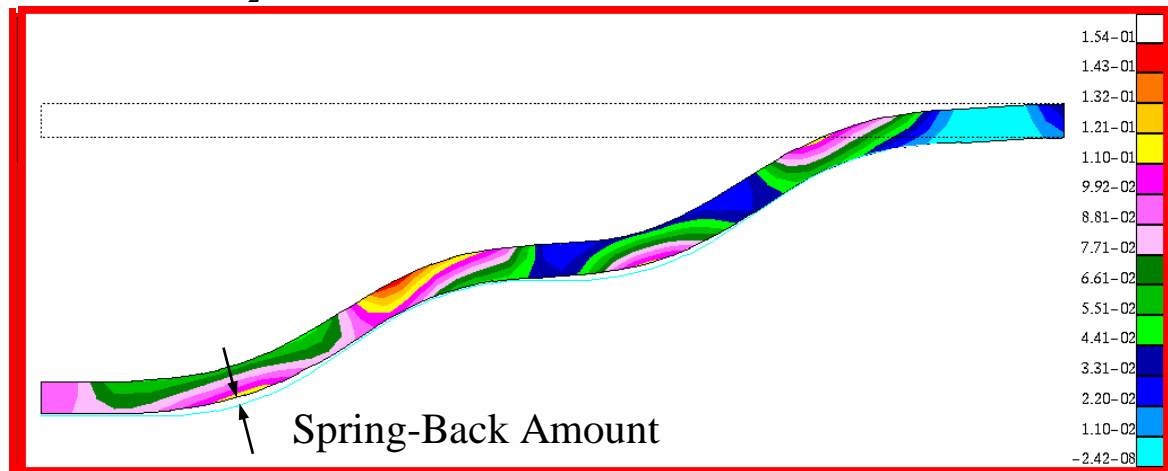
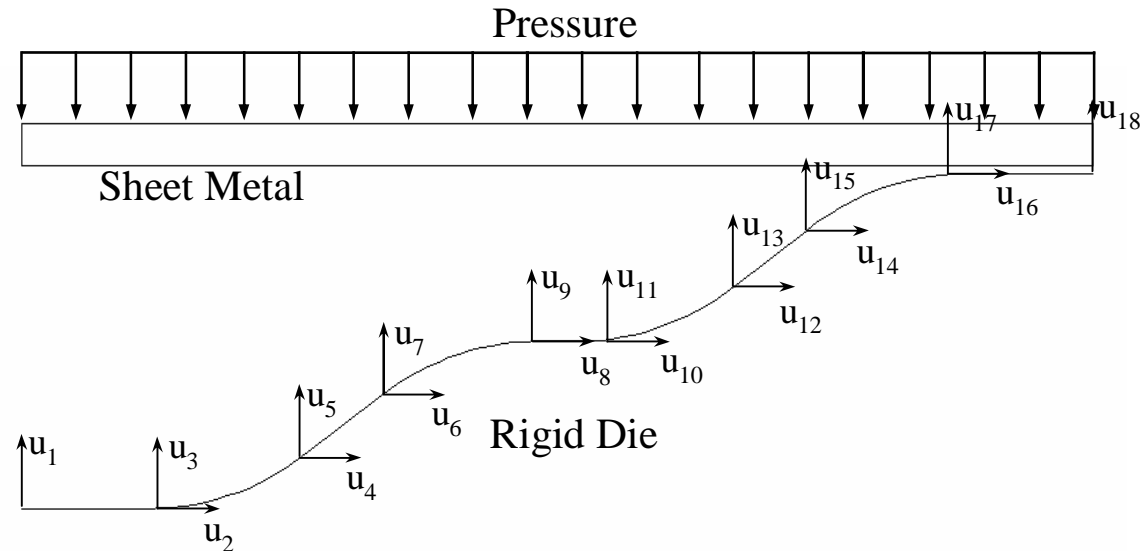
Effective Plastic Strain Von Mises Stress

DESIGN OPTIMIZATION HISTORY



Response Analysis : 37
Sensitivity Analysis : 18

SHEET METAL STAMPING PROBLEM



DESIGN OPTIMIZATION

Optimization of Sheet Metal Stamping (Static)

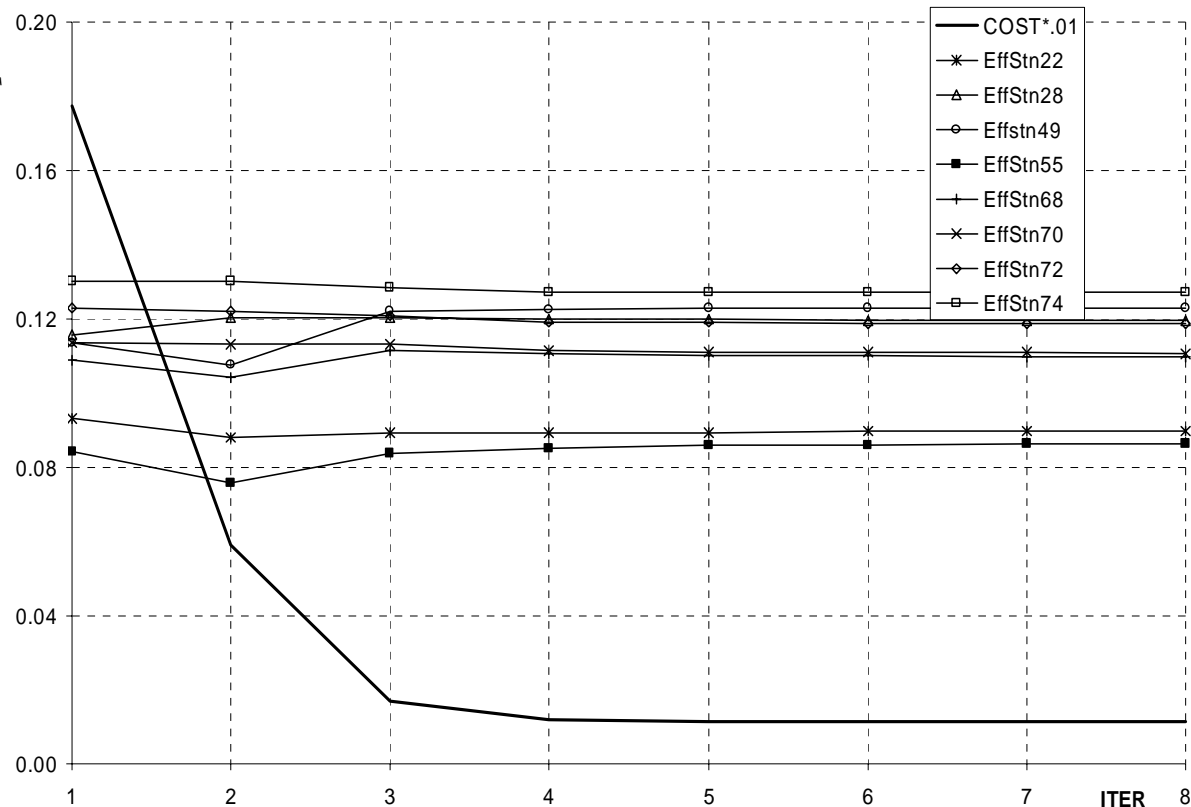
$$\text{MIN } G = \int_{\Gamma} \|\pi(\mathbf{x}) - \mathbf{x}\|^2 d\Gamma$$

$$\text{S.T. } \hat{e}^p_i \leq 0.13$$

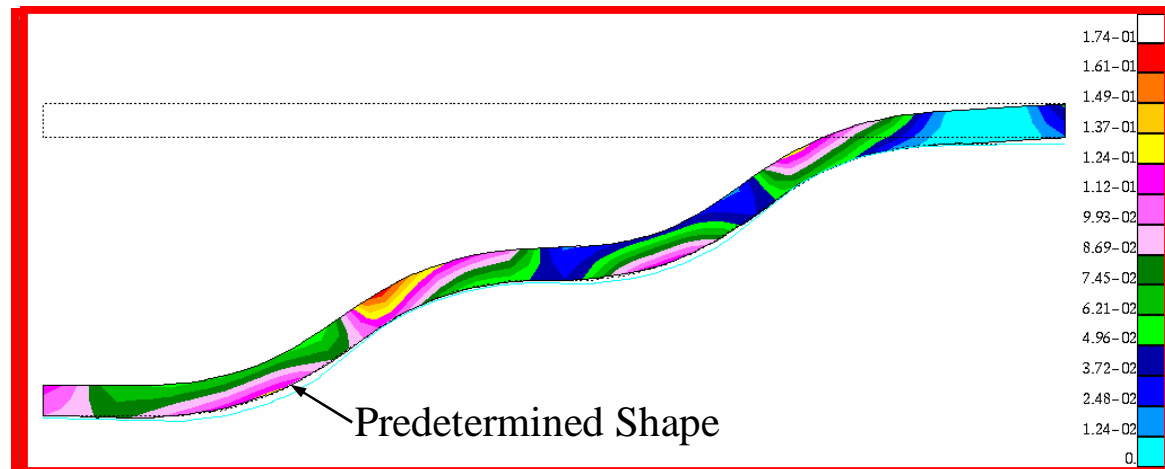
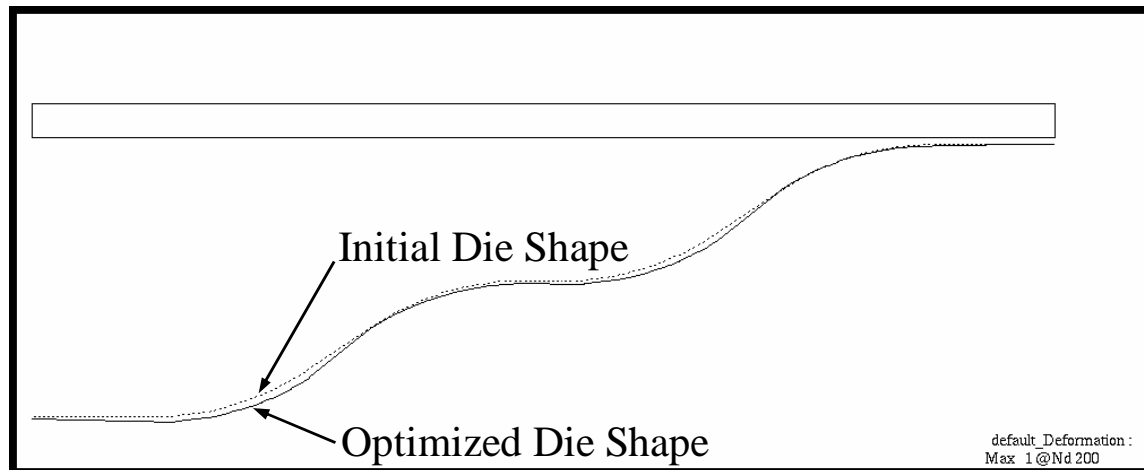
$$-2.0 \leq u_i \leq 2.0$$

Response Analysis : 24

Sensitivity Analysis : 8



OPTIMUM DIE SHAPE



CONCLUSIONS

- ⇒ An efficient and accurate nonlinear DSA is proposed for path-dependent structural problems.
- ⇒ Sensitivity equation always uses the same tangent stiffness matrix as the response analysis at each converged configuration.
- ⇒ Path-dependency of DSA is from the intermediate configuration and internal plastic variables as well as frictional contact effect.
- ⇒ Intermediate configuration is required to be specified for DSA purpose such that the incremental plastic rotation is ignored.